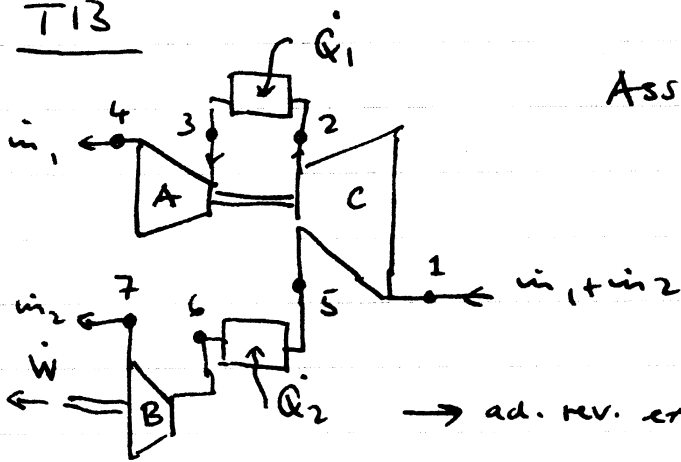


T13

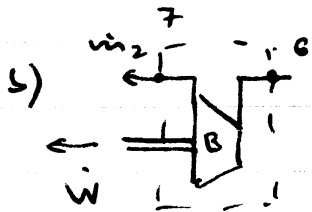
16. Unified Fall 07



Assume: - ideal turbomachinery
 - adiabatic flow except in combustor
 - ideal gas with const. spec. heats
 - neglect KE, PE effects
 → ad. rev. exp. in turbines, ad. rev. compr. in compr.

a)

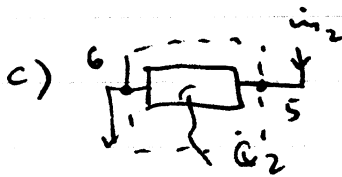
$P_1 = 1 \text{ bar}$	$P_2 = 5 \text{ bar}$	$P_3 = 5 \text{ bar}$	$P_4 = 1 \text{ bar}$
$T_1 = 300 \text{ K}$	$T_2 = T_1 \left(\frac{P_2}{P_1}\right)^{\frac{\gamma-1}{\gamma}} = 475 \text{ K}$	$T_3 = 1000 \text{ K}$	$T_4 = T_3 \left(\frac{P_4}{P_3}\right)^{\frac{\gamma-1}{\gamma}} = 631 \text{ K}$
	$P_5 = 5 \text{ bar}$	$P_6 = 5 \text{ bar}$	$P_7 = 1 \text{ bar}$
	$T_5 = 475 \text{ K}$	$T_6 = 1000 \text{ K}$	$T_7 = 631 \text{ K}$



$$\frac{dE_{cv}}{dt} = \sum \dot{Q} + \sum \dot{W} + \sum \dot{m} h$$

$$0 = -\dot{W} + \dot{m}_2 h_6 - \dot{m}_2 h_7$$

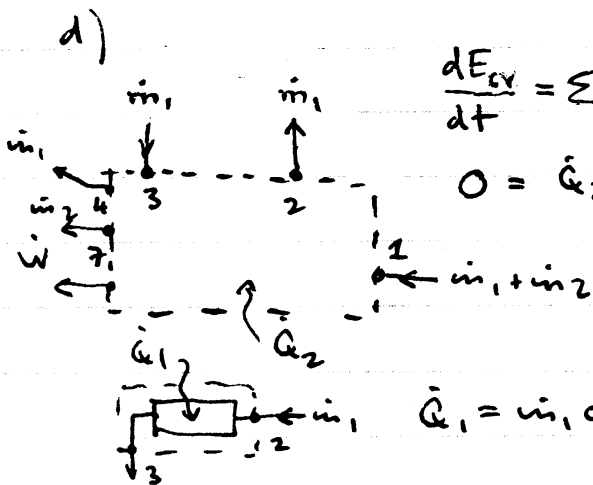
$$\dot{m}_2 = \frac{\dot{W}}{c_p(T_6 - T_7)} = 4.65 \text{ kg/s}$$



$$\frac{dE_{cv}}{dt} = \sum \dot{Q} + \sum \dot{W} + \sum \dot{m} h$$

$$0 = \dot{Q}_2 + \dot{m}_2 h_5 - \dot{m}_2 h_6$$

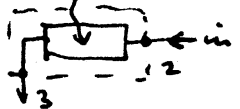
$$\dot{Q}_2 = \dot{m}_2 c_p (T_6 - T_5) = 2.14 \text{ MW}$$



$$\frac{dE_{cv}}{dt} = \sum \dot{Q} + \sum \dot{W} + \sum \dot{m} h$$

$$0 = \dot{Q}_2 - \dot{W} + (\dot{m}_1 + \dot{m}_2) h_1 - \dot{m}_1 h_2 + \dot{m}_1 h_3 - \dot{m}_1 h_4 - \dot{m}_2 h_7$$

$$\dot{m}_1 = \frac{\dot{W} - \dot{Q}_2 + \dot{m}_2 c_p (T_7 - T_1)}{c_p (T_1 - T_2 + T_3 - T_4)} = 3.63 \text{ kg/s}$$



$$\dot{Q}_1 = \dot{m}_1 c_p (T_3 - T_2)$$

$$\dot{Q}_1 = 1.91 \text{ MW}$$

e)

$$\eta_{th} = \frac{\dot{W}}{\dot{Q}_1 + \dot{Q}_2}$$

$$\eta_{th} = 0.37$$